



APISAT 2019

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ENGINEERS
AUSTRALIA





Electric Systems

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Impact of Propulsion Technology Levels on the Sizing and Energy Consumption for Serial Hybrid-Electric General Aviation Aircraft



- Goal:
- Identify most important technology parameters
 - Quantify possible improvements
→ Technological thresholds

Means: Parameter variation studies for
2 aircraft starting from
3 technology levels

Agenda

1. Initial Sizing for Hybrid-Electric Aircraft
2. Baseline Aircraft and Technology Levels
3. Parameter Studies E-Motor P/W & Battery Energy Density
4. Conclusion

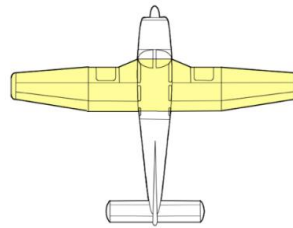
1. Initial Sizing for Hybrid-Electric Aircraft

- Initial Sizing = First calculation of mass, wing area, thrust/power

MTOM



Wing Area



Thrust / Power



- Conventional methods e.g. Raymer, Roskam, Gudmundsson

Fuel burning ICE
 → A/C loses weight
 → Breguet range formula

EM powered from batteries
 → A/C does not lose weight
 → Modified Breguet formula

Either one



Conv. methods

Both



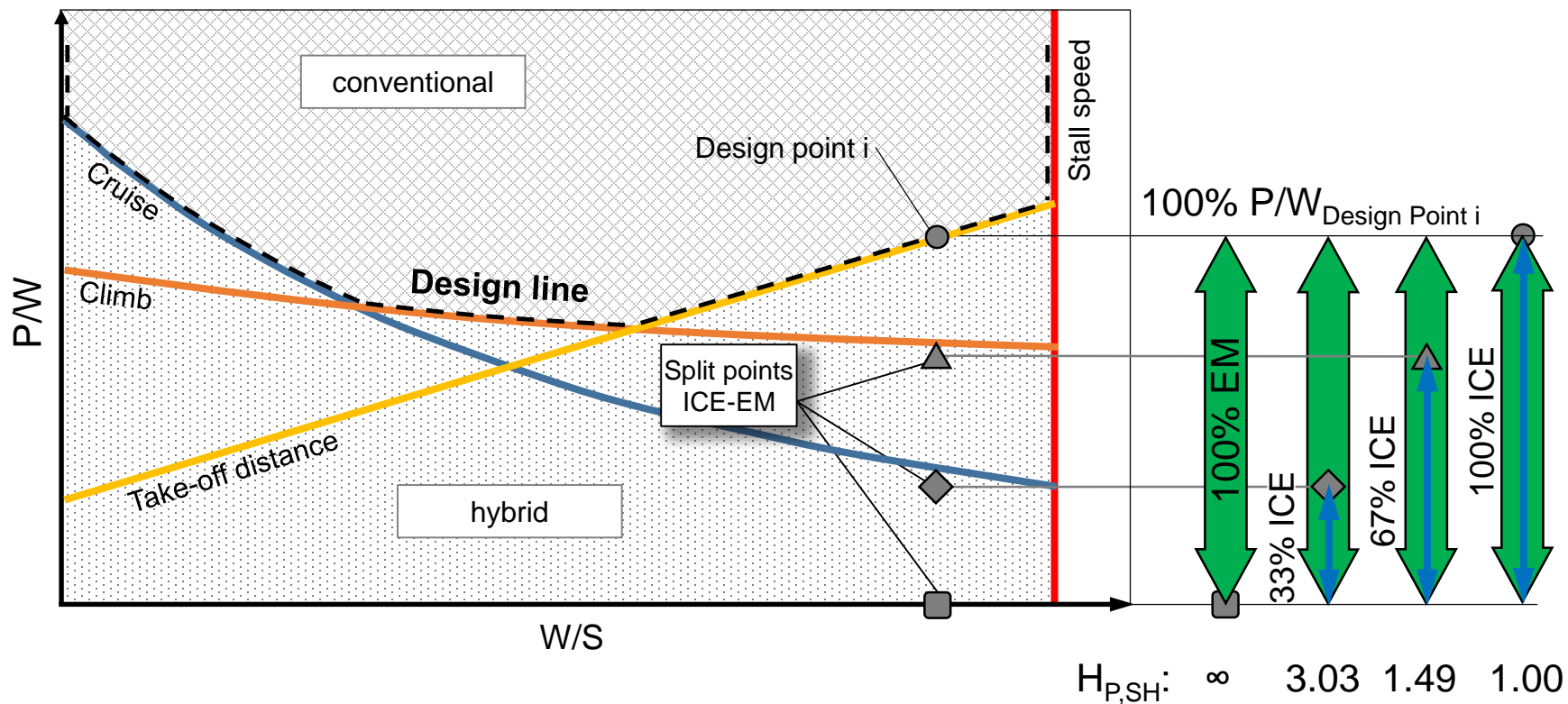
Other method needed

1. Initial Sizing for Hybrid-Electric Aircraft

- Sizing method for hybrid-electric GA aircraft from FH Aachen UAS, described by (Finger 2018)
- 2 key points of this method
 1. Energy based approach for iteration of weights & new empty-mass-fractions
 2. Definition of Degrees of Hybridization using the Matching Diagram

1. Initial Sizing for Hybrid-Electric Aircraft

Definition of Degrees of Hybridization for SERIAL hybrid using the Matching Diagram



Degree of Hybridization
of Power

$$H_{P,SH} = \frac{P_{E-Motor}}{P_{ICE}}$$

Degree of Hybridization
of Energy

$$H_{E,mean} = \frac{E_{non-consumable}}{E_{total}}$$

1. Initial Sizing for Hybrid-Electric Aircraft

- 2 assessment criteria and design objectives are used

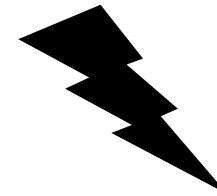
Maximum Take-Off Mass (MTOM)



- Most common
- Correlates to cost

and

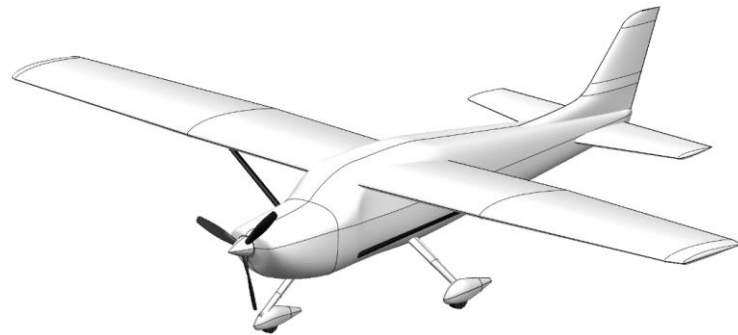
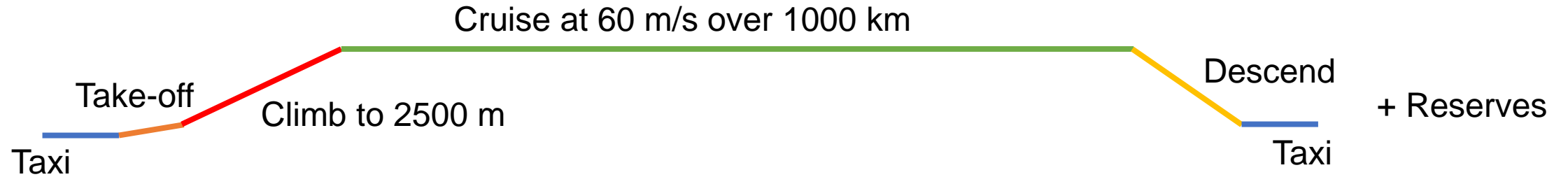
Primary Energy Consumption (PEC)



- Total expended energy to provide fuel / electricity
- Factors from Germany 2016
- Fossil fuel = 1.1 (refinement process from raw oil)
- Electricity = 2.8 (electricity mix, coal, gas, ...)

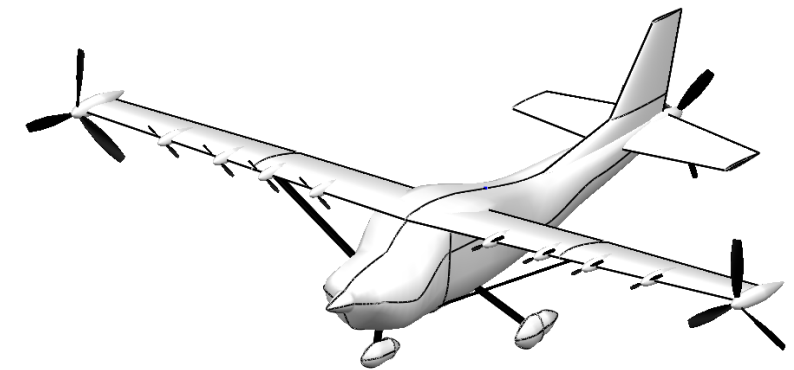
2. Baseline Aircraft and Technology Levels

- 2 serial hybrid-electric baseline aircraft with same mission



Aircraft #1

	Aircraft #1 Non-DEP	Aircraft #2 DEP
Payload [kg]	340	340
c_{D0} [counts]	290	250
$c_{L,max}$ [-]	2	4
$c_{L,TO}$ [-]	1.2	2



Aircraft #2

2. Baseline Aircraft and Technology Levels

- 3 technology levels

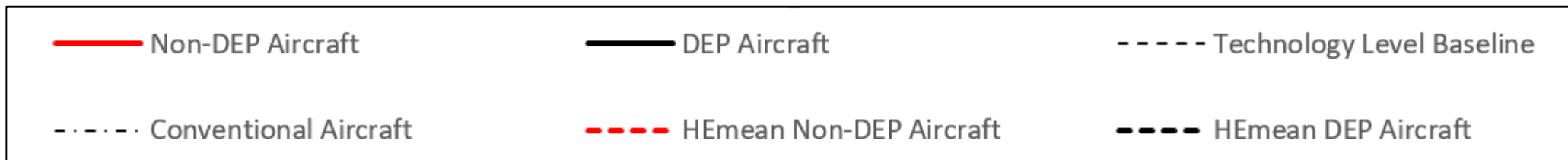
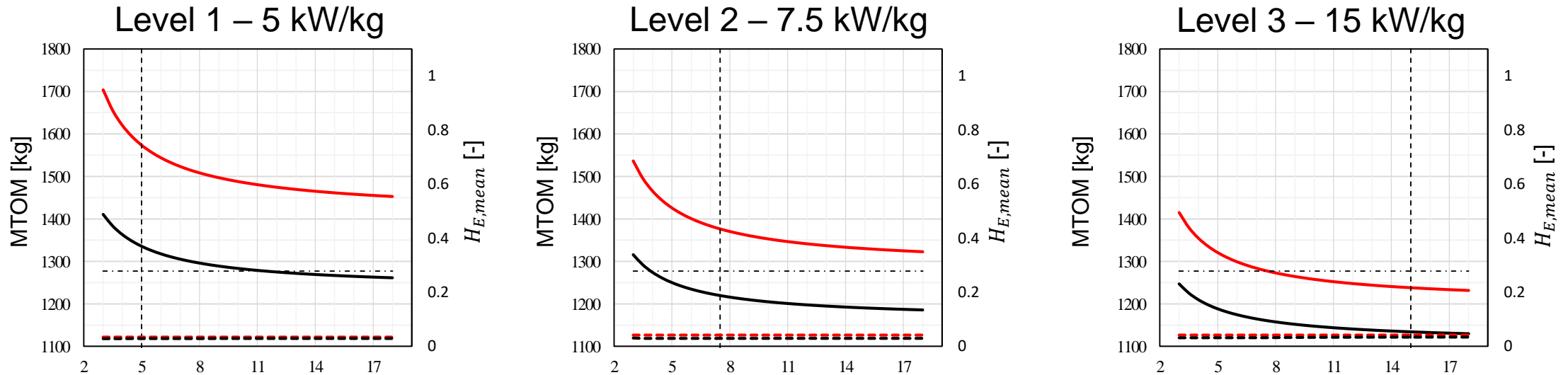
	Level 1 – today	Level 2 – 2035	Level 3 – 2050
Battery Specific Energy [Wh/kg]	250	400	1000
Battery Specific Power [kW/kg]	2.5	4	10
E-Motor Specific Power [kW/kg]	5	7.5	15
E-Systems Efficiency [-]	0.91	0.94	0.96

↳ = E-Motor Efficiency x Battery Efficiency x Controller Efficiency

- Further parameters that were changed with the technology levels:
ICE Specific Power, Primary Energy Factors

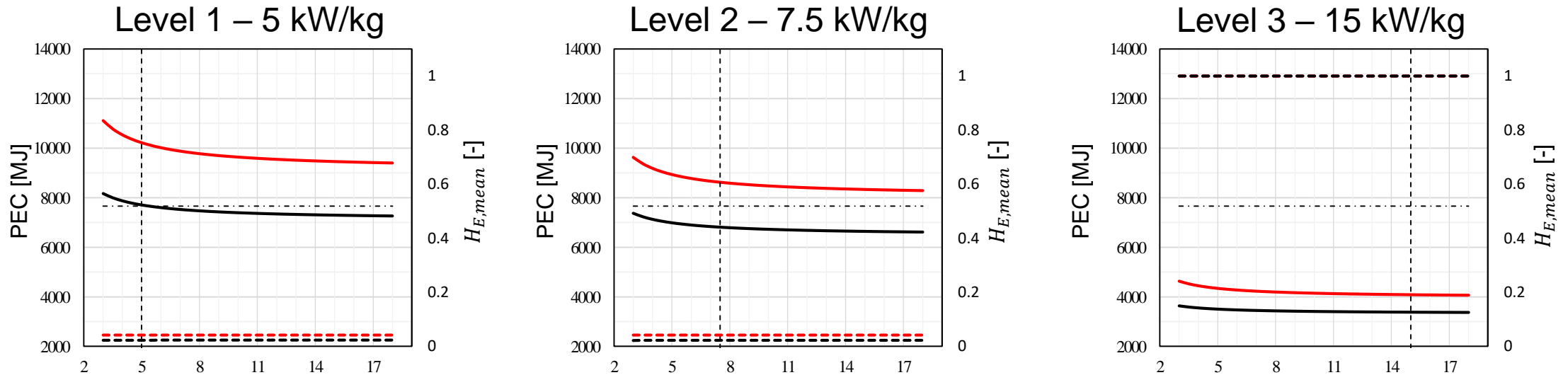
3. Selected Parameter Studies

- E-Motor Specific Power varied from 3 - 18 kW/kg (lowest MTOM)



3. Selected Parameter Studies

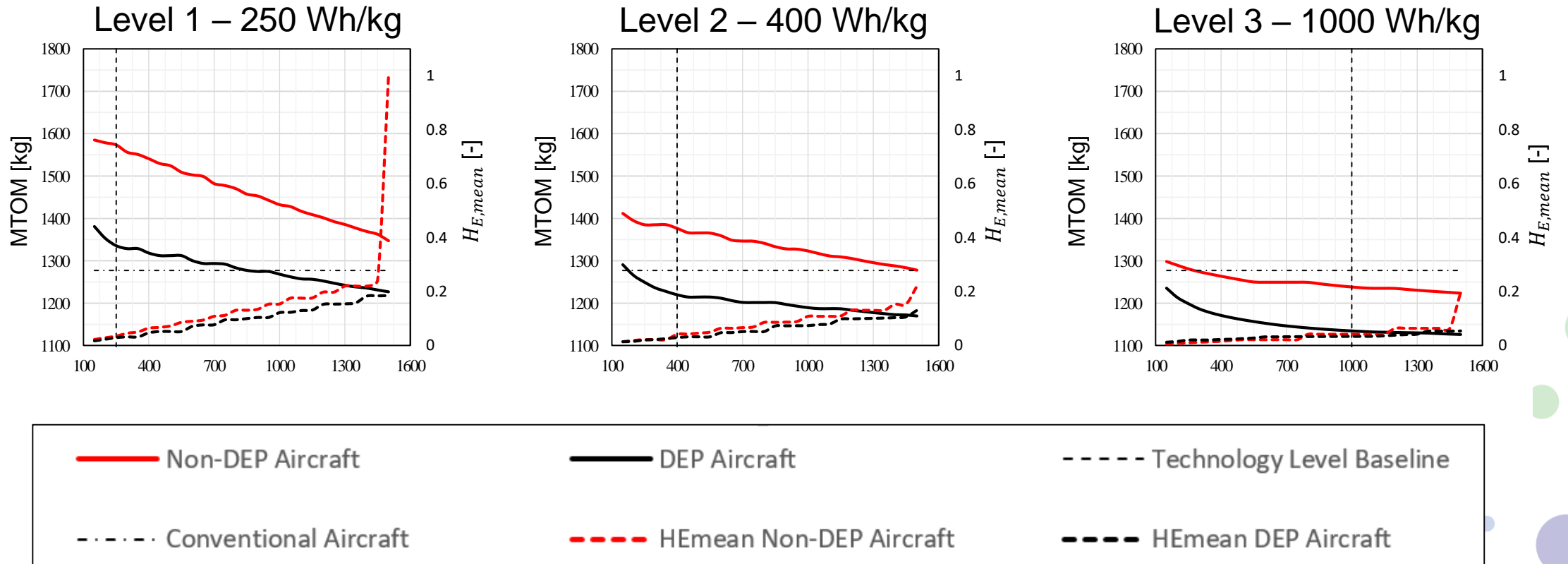
- E-Motor Specific Power varied from 3 - 18 kW/kg (lowest PEC)



- Results:
- MTOM and PEC generally decline with increasing P/W
 - Gradient also declines
 - Technological threshold at around 5 – 8 kW/kg

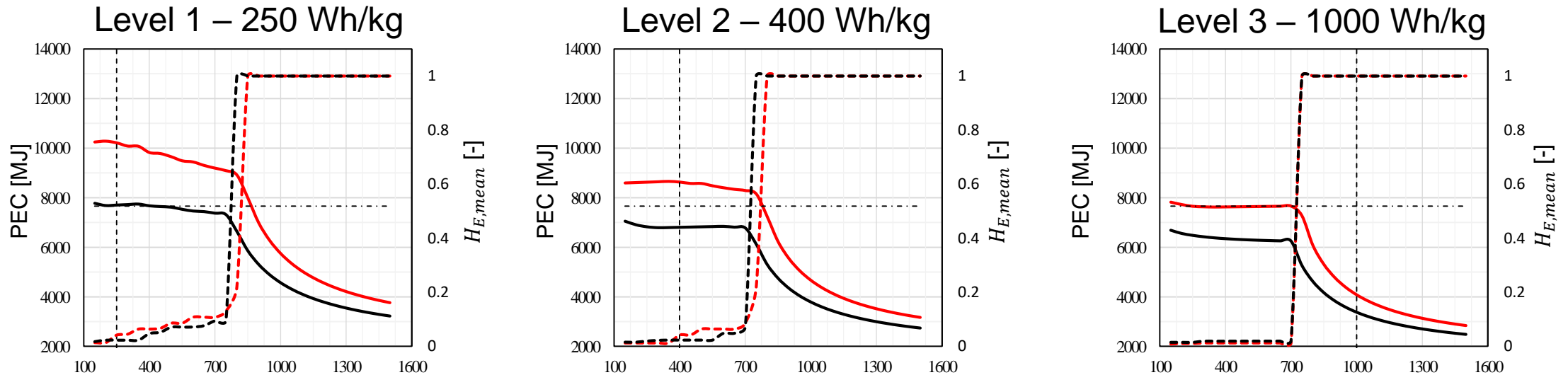
3. Selected Parameter Studies

- Battery Specific Energy varied from 150 - 1500 Wh/kg (lowest MTOM)



3. Selected Parameter Studies

- Battery Specific Energy varied from 150 - 1500 Wh/kg (lowest PEC)



- Results:
- MTOM nearly linear decline, PEC linear/flat then drops
 - Drop is switch from hybrid to fully electric aircraft
- Long-term fully electric aircraft superior to hybrid for lowest PEC

4. Conclusion

- Similar trends for both baseline aircraft and design objectives (MTOM and PEC) except for Battery Specific Energy
- E-Motor technological threshold around 5-8 kW/kg
- Fully electric aircraft best solution for lowest PEC when 700 Wh/kg batteries are available

References

- Finger, DF, Braun, C & Bil, C 2018, 'An Initial Sizing Methodology for Hybrid-Electric Light Aircraft', in *2018 Aviation Technology, Integration, and Operations Conference*, Atlanta, Georgia.
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- Roskam, J 1990, *Airplane design*, Airplane cost estimation: design, development, manufacturing and operating, Roskam Aviation and Engineering Corp., Ottawa, Kan.



**Thank you for your attention.
Questions?**